Submit original and one copy without letter of transmittal to the cognizant Bureau of Aeronautics Maintenance Representative and one copy to the following activities: (1) NAVAVNSAFEACTY NAS Norfolk, (2) Controlling Custodian, (3) OKR, (4) Reporting Custodian, (5) Log Rook, (6) Controlling Activity (if ferry mircraft), (7) ComNAB, (if ferry or other transient mircraft and a ComNAB is concerned), and (8) Ferry Squadron, (if concerned). Commanding Officer, MAS North Island, San Diego, Calif. 8-66 1 Feb. 196 92135 Name of Contrast o BUNEPSPLETEADREP PAC (a) LATEST ISSUE OF BUARR INSTRUCTION NAVAER 00.38 DATE OF ACCIDENT OR DAMAGE 12 Dec. 1965 MAS MORES BUNEPS (FR) SERVICE TOUR P-4B 2 SEE BOTE #1 NTHS THIS SERVICE TOUR 151001 ENGINE DATA 17 PORT STARBOARD STATUS (Flyable, non-flyable) URS THIS MONTH Mon-Flyable 0.0 BUREAU NUMBER TIME ON ENGINE(S) 10 June 1963 538.4 TOTAL TENES PAR AS SINCE NEW TIME ON ENGINE(S) SINCE OVERHAUL

SEE ENCLOSURE (1)

902.7

TOTAL DIRECT 11,800 MANHOURS TOTAL MANHOUR COST 109,032.00 NON-REPAIRABLE ASSEMBLIES COST \$ 376,539.00 MATERIAL 12,110.00 SHIPPING OR TRANS PORTING TO GER TOTAL COST 497.681.00

ESTIMATED COSTS OF LABOR AND MATERIAL NEC ESSARY TO PLACE AFRICAFT IN COMPLETE SERVICEABLE CONDITION

Crash Damage Overheul

The following preparatory work shall be accomplished by the operating unit prior to acceptance by the Overhaul and Hepair Depart-

- a. Remove ammunition, pyrotechnics and ejection seat charge.
- b. Remove spare and loose gear.
- c. Preserve engine(s).

RECOMMENDED DISPOSITION

- d. Drain and purge fuel cells and disconnect battery.
- e. Bring log books up-to-date and deliver with sircraft. f. Inventory aircraft and retain copy for reference.

ESTIMATED INDUCTION DATE FOR REPAIR (Contingent on eveilability of exterial) ESTIMATED COMPLETION DATE SIGNATURE OF PLANNER AND ESTIMATOR NAVAVNSAFEACTY NAS Norfolk Controlling Custodian BUNES (FFM-25)

O. H. HICOLAI

Reporting Custodian

en dest Mill AFS Utch (Attn:

Bury Intere Linion Office R. J. MAZELA

By direction

F-4B, BUNO 151001

- 1. The RH center wing suffered severe structural damage to the following:
 - a. Main spar and intermediate rib are broken at their intersection.
 - b. Rear lower spar cap is broken below main landing gear trunnion.
- c. Aft main landing gear actuator rib is buckled near cylinder attach point.
- d. Upper main wing skin is buckled over intermediate rib and main spar.
- e. Bottom trailing edge skin is split 20 inches and pushed down just aft of main spar outboard of intermediate rib.
- f. Bottom skin is torn 3" X 9" along inboard side of forward main landing gear rib with resultant damage to adjacent rib flange. The skin is also broken and an 8 inch crack carries forward from immediately below forward main landing gear trunnion.
 - g. Honeycomb at outboard end of main landing gear well is torn open.
- Entire fuselage nose structure is severely damaged thru F.S. 77.00 and the lower skin and structure outboard of L/R keels to F.S. 95.50.
- 3. All equipment forward of F.S. 77.00 is missing.
- 4. L/R refrigeration units received minor damage.
- 5. Nose landing gear installation and door are damaged beyond repair.
- 6. L/R main landing gear installations (except RH actuator) are missing.
- 7. RH main landing gear door has minor skin buckle.
- 8. RH inner wing LE flap is missing and attaching wing hinge half is damaged.
- 9. RH inboard LE flap is buckled and torn along entire length.
- 10. RH outer wing panel LE flap has 3 feet of inboard end torn away.

ENCLOSURE (1)

- 11. Bottom skim of RH outer wing panel is gouged (approximately 12 inches long) adjacent to feld rib. The fold rib fairing is partially torn away.
- 12. Hi outer wing panel honeycomb TE has minor dents.
- 13. RE TE flap actuator is broken and the flap is buckled and torn beyond repair.
- 14. RH aileron honeycomb TE is partially torn away.
- 15. LH TE flap honeycomb is torn.
- 16. LH aileron bottom skin is punctured and honeycomb is torn.
- 17. LH speed brake bottom skin has a minor puncture.
- 18. LH main landing gear inboard door is buckled beyond repair.
- 19 IN inner wing flap skin is torn at inboard end.
- 20. Doors 82L, 81L, 82R and 83L are torn and buckled beyond repair.
- 21. Door 83R has slight skin buckle.
 - NOTS: 1. The L/R engine installations were not received with the aircraft.
 - 2. OPHAV INSTR. P3750.6E, Estimate: 6,000 manhours.

31RCBidef 3750-3

AIR MAIL UNCLASSIFIED 3750-3 14 December 1965

To: Commander
U. S. Naval Aviation Safety Center

Subj: Report of Aircraft Mishap

Direct Mnemy Action

1. F-4B, 151001, VNFA-323

2. 12 December 1965, 1336H, Runway 35 Da Nang AB

(DAY) LONDIN

3. Bravo. Mose and right main landing gears collapsed with attendant damage to attachment fittings. Right wing leading edge flaps torn away. Trailing edge flaps and ailerons on both wings punctured and torn. Functures and dents in undersides of wings and fuselage. All pylons and had/Tak racks (except left outboard MaR) torn away. Radome and fuselage structure as far back as pilots instrument panel distorted and torn. Damage incurred in the mishap precludes determination of the actual extent of battle damage.

- 4. The pilot experienced a double generator failure on pullout from a bombing run on a target some twenty-five miles south of Da Nang. He secured both generator switches, extended the RAT, and, concluding battle damage, turned toward Da Nang for immediate recovery. Intermittent left generator operation was regained on the return trip; this generator then out out twice again, Right generator operation could not be regained at any time. Two other emergencies were in the pattern, plus normal traffic, when the pilot arrived at Da Nang, The flight leader had joined on the distressed aircraft by this time but, seeing nothing amiss, and because of congestion on tower frequency due to the many emergencies in progress made no comments. The tower attempted to clear the pattern but some ill-considered advice to the tower from an unidentified aircraft set up a near miss mid-air collision situation at the runway threshold. The pilot involved in this mishap was forced to execute a last second wave off. He made an immediate turn down wind and, in his anxiety to get aboard, occasioned by his own emergency status coupled with other emergency status aircraft still in the pattern, made a hard landing at near maximum "flared" landing gross weight. This caused the left tire to blow and fractured the wheel rim. Pieces of the rim sheared to hydraulic and preumatic lines to the brake. This caused utility hydraulic failure reducing effectiveness of the right brake and eliminating nose gear steering. Full right brake, rudder and spoiler were insufficient to return the aircraft to runway heading and it ran off the left side of the runway after only approximately 1,500' of roll.
- 5. This pilot was well experienced with over 325 hours in the F-4B. He had flown six "combat" missions but this was the first where he had personally observed ground fire. His immediate reaction was that his aircraft had been hit in view of the fact that three other F-4B aircraft had been struck by ground fire in that immediate area within the previous few days. The recovery from the bombing run and the double generator failure occurred at approximately 1200' AGL in a solid overcast, thus somewhat contributing to the less than thorough, calm analysis of the situation. The simultaneous generator

SPECIAL HANDLING RE JURNED in accordance with para 66, Upway Inst 13750, 66

COPY

failure and erratic performance of the left generator are difficult to explain However, such failures, without the added stress of combat, have led to precipitate action in the past (VAF(as)-531 AAR 1-62 refers). In this mishan

6. Claude D. HILLIS (b) (6) 7307 "F" plus 1 "G".

talke to some time.

aw Some.

2

COPY TOP NASC (Orig &1) CG, First MAN (1) COMNAVAINPAC (1)

CMC (AAP) (1) CO, MAG-11 (1) COMPAIRML STPAC (1)

CG, FMFPac (1) CO, WEG-1 (1)

File (2)

COMMANDING OFFICER
VIFA-323, Mag-11, 1stM.w, III haf, FMFPac
FPO, San Francisco, California 96601